

# FAILURE MODES AND EFFECTS ANALYSIS

ASSY Nomenclature: RMS IFM DEC A/T  
ASSY PN: SIE3310304-304

SYSTEM: REMOTE MANIPULATOR SYSTEM  
SUBSYSTEM: RMS IN-FLIGHT MAINTENANCE DEC KIT

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NAME AND DRAWING	FUNCTION	FAILURE MODE AND CAUSE	MISSION PHASE	FAILURE EFFECT ON			FAILURE DETECTION	CORRECTIVE ACTION TIME AVAILABLE/ TIME REQUIRED	CRIT. H/F	HAZARDS/REMARK
				END ITEM	MISSION	CREW/VEHICLE				
EE COMMAND SIGNAL CIRCUIT, QTY. 1, (SET OF RESISTOR [R2], ZENER DIODES [D3, D4], AND CAPACITOR [C3]) SIE3310307	Provides ref. signals for end effector commands	6090 Mode: Zener diode D4 fails short	Orbit	Loss of ability to capture or rigidize EE. Capture command will give release command. Rigidize command will give desolidize command. Release and desolidize commands will be normal	<u>Worst Case</u> Unexpected payload motion. Incomplete rigidization. Crew action required.	Incomplete rigidization resulting in unexpected payload motion which could cause vehicle damage	Operator detects EE off nominal operation. EE 1B's on RMS D&C panel may change	Release payload using backup release or jettison arm/payload combination immediate/immediate	1/1	Crew action required to prevent motor burnout
		Cause: <ul style="list-style-type: none"><li>* structural failure</li><li>* mechanical stress</li><li>* vibration</li><li>* electrical stress</li><li>* resistor R2 fails short</li></ul>	Orbit	Loss of desolidize and release commands. Capture and rigidize commands operate normally		Subsequent failure may result in inability to release payload	Operator detects payload has not been released. EE 1B's on RMS D&C panel do not change	Release payload using backup release or jettison arm/payload combination immediate/hours	3/1A	<u>Redundancy Screen:</u> <ul style="list-style-type: none"><li>A - N/A</li><li>B - N/A</li><li>C - N/A</li></ul>
EE COMMAND SIGNAL CIRCUIT, QTY. 1, (SET OF RESISTOR [R2], ZENER DIODES [D3, D4], AND CAPACITOR [C3]) SIE3310307	Provides ref. signals for end effector commands	7090 Mode: Zener diode D4 fails open, fails to conduct		Cause: <ul style="list-style-type: none"><li>* structural failure</li><li>* mechanical stress</li><li>* vibration</li><li>* electrical stress</li><li>* resistor R2 fails short</li></ul>	<u>Worst Case</u> Subsequent failure may result in inability to release payload	<u>Remaining</u> Backup mode. RMS jettison				<u>Redundancy Screen:</u> <ul style="list-style-type: none"><li>A - Pass</li><li>B - Pass</li><li>C - Pass</li></ul>
		Cause: <ul style="list-style-type: none"><li>* structural failure</li><li>* mechanical stress</li><li>* vibration</li><li>* electrical stress</li><li>* resistor R2 fails short</li></ul>								

SOLODOCK ATTACHMENT  
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DATE: 9/90

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